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# Engine, Transmission Failures Lead Causes of Accidents in U.S. Helicopter Logging Operations

More than half the accidents in U.S. helicopter logging operations in 1983 through 1999 involved failures of engines or transmission systems. Metal fatigue and external-load problems also were major causes of accidents.

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Helicopter operations in support of logging activities in the United States typically are conducted in remote areas where rugged terrain and adverse flight conditions present unique risks for flight crewmembers and ground crewmembers. In 1983 through 1999, 83 accidents occurred during helicopter logging (heli-logging) operations; 23 accidents were fatal, and 29 crewmembers were killed.

To identify accident causes and potential methods of improving U.S. heli-logging safety, the author conducted a study that included the following:

- Analysis of U.S. National Transportation Safety Board (NTSB) reports on U.S. heli-logging accidents from January 1983 through December 1999 (see "Helicopter Accidents During U.S. Helicopter Logging Operations, 1983–1999," page 5), interviews of some accident investigators and some witnesses, and inspections of some accident sites;
- Examination of training documents and training courses;
- Examination of operations specifications and operations manuals;
- Observation of several heli-logging operations; and,
- Examination of heli-logging helicopters.



The study produced the following major findings:

- All the accident helicopters were carrying loaded external lines or unloaded external lines;
- Most accidents (88 percent) occurred while helicopters were being maneuvered in work areas; and,
- Mechanical failure caused more than half (57 percent) of the accidents.

Figure 1 (page 2) shows that 73 accidents involved helicopters being maneuvered in work areas; 20

accidents resulted in fatalities, and 22 accidents resulted in serious injuries.

Figure 2 (page 2) shows that 46 maneuvering-phase accidents were caused by mechanical failure and that 27 maneuvering-phase accidents were caused by human error (other than human error during maintenance).

Maneuvering typically is conducted for extended periods at low altitudes over steep, mountainous terrain. Helicopters often are flown close to terrain and obstacles that limit maneuvering room. High density altitude often reduces hovering capability. Maneuvering at low altitudes and slow airspeeds reduces the probability of a successful autorotative landing following a powerplant failure.





#### Continued from page 1

Seven accidents occurred during the en route phase of flight that is, while helicopters were being flown between staging/ refueling areas and work areas. Three en route accidents resulted in fatalities. Six en route accidents involved human error.

Engine failures were involved in 22 accidents, including 15 accidents involving mechanical failure, five accidents involving fuel starvation or fuel exhaustion, and two accidents involving fuel contamination. All the engine-failure accidents occurred while the helicopters were being flown at low altitudes and/or low airspeeds. Four accidents resulted in fatalities; eight accidents resulted in serious injuries.

Eleven accidents involved metal fatigue that caused mechanical failure of tail-rotor drive-shaft systems. Six accidents involved failure of the 42-degree gearboxes in Bell UH-1 (and related) series helicopters. The NTSB report on the Bell TH-1L accident July 12, 1988, in Walterboro, South Carolina, said that, although the U.S. Federal Aviation Administration (FAA) had certified the military-surplus helicopter for civilian external-load operations, the helicopter was not designed for repeated heavy lifting. The report said that the accident helicopter had accumulated more than 13,000 lift cycles in 380 heli-logging flight hours.

Seven accidents occurred when external lines became entangled with tail-rotor systems.

Three accidents involved fatigue-induced failures of vertical stabilizers on Garlick UH-1/TH-1 helicopters. [Garlick Helicopters refurbishes Bell UH-1/TH-1 helicopters for civilian use. UH-1 (utility) series helicopters and TH-1 (training) series helicopters are military versions of the Bell 204 and 205.]

Four accidents involved main-rotor system failures. One accident occurred when the main-rotor system separated from a Southern Aero UH-1B helicopter in Skykomish, Washington, on Aug. 27, 1997. NTSB said, in its final report, that the probable cause of the accident was a main-rotor mast fatigue fracture.

Eleven accident reports said that the pilots attempted to release their external loads after experiencing problems in flight. External loads are attached to cargo hooks on helicopter external lines. The cargo hooks must be designed to enable pilots to use either electrical systems or manual systems to release external loads in normal operations and emergency situations. The release mechanisms operated properly in eight accidents and malfunctioned in three accidents.

Four accidents involved ground crewmembers who were struck and killed by external loads that were released accidentally or because of malfunctions of devices containing the loads.

Twelve accidents occurred when external loads or unloaded external lines struck trees or the ground while the helicopters

were being maneuvered. Seven accidents occurred when helicopters struck obstacles (typically trees) while being maneuvered; five accidents occurred when tail rotors struck trees. All the tree strikes occurred while the pilots were conducting vertical-reference work, which typically involves leaning outside the cockpit and looking down.

Improper coordination between flight crewmembers or between flight crewmembers and ground crewmembers was cited in 12 accident reports. Two accidents involved relatively inexperienced pilots. Two accidents involved relatively inexperienced ground crewmembers.

Heli-logging activities are conducted year-round. Figure 3 (page 3) shows the monthly distribution of heli-logging accidents during the study period.

Figure 4 (page 3) shows the distribution by state of heli-logging accidents. Two-thirds of the accidents occurred in Alaska, Oregon and Washington.

The remote areas in which some heli-logging operations are conducted are far from company headquarters and maintenance facilities. Poor maintenance oversight was cited in seven accident reports; five of the accidents involved improper installation of retaining nuts and bolts.

The rugged terrain over which most heli-logging operations are conducted provides limited choices of landing sites for pilots who experience power problems or control problems. Unsuitable terrain for (forced) landings was cited in 53 accident reports. Twenty-seven accidents involved pilots who attempted to conduct autorotative flight to suitable landing sites but had to land elsewhere because of insufficient altitude and/or airspeed. Seventeen accident reports said that the pilots attempted to "stretch glides" to suitable landing sites and allowed main-rotor speed to decay.

Four accident helicopters descended into water in rivers or swamps. All the water-accident reports said that crewmembers had difficulty exiting the helicopters. Some crewmembers were hindered by injuries, disorientation, in-rushing water or obstacles. None of the crewmembers was wearing a personal flotation device. ♦

### About the Author

Patrick R. Veillette, a professional pilot with more than 11,500 flight hours, is a Boeing 727 flight engineer for a U.S. air carrier. Veillette earned a bachelor's degree in aeronautical engineering at the U.S. Air Force Academy and a doctorate in civil engineering at the University of Utah. He has conducted research projects on cockpit automation and human error in high-risk environments. Veillette has an airline transport pilot certificate and is a former U.S. Federal Aviation Administration designated pilot examiner.

## Appendix Helicopter Accidents During U.S. Helicopter Logging Operations, 1983–1999

Date	Location	Helicopter Type	Helicopter Damage	Injuries
Aug. 20, 1983	Shaver Lake, California	Boeing Vertol BV-107	destroyed	2 fatal
A ground crewmen a log that weighed rotor-drive system service hours sind	mber heard noises and saw the rod d between 8,000 pounds and 9,00 had failed at a fatigue crack that o e overhaul. The report said that t	tor blades fragment as the pilot 00 pounds (3,629 kilograms ar originated at a rivet hole. The sh he quality of the surface finish	began to fly the helicopter awa d 4,082 kilograms). The no. 3 aft had accumulated 17,492 so on the rivet holes was unsatisf	y from the pickup area with synchronizing shaft in the ervice hours, including 435 factory.
May 10, 1984	Clermont, Florida	Bell UH-1B	minor	1 fatal, 1 serious, 1 none
The pilot was not a helicopter, and bo malfunction. The r	aware that two ground crewmemb oth crewmembers fell to the grour report said that similar incidents h	ers were holding the external lind. Examination of the externa ave occurred with civilian and	ne. The external line inadverte I-line attachment hook showed military helicopters equipped v	ntly was released from the d no evidence of failure or vith hooks.
Dec. 10, 1984	Carrabelle, Florida	Sikorsky UH-34J	substantial	none
The pilot was flyin engine operated in landing in a creek	g the helicopter in a hover at aboutermittently and that black smoke . The main-rotor blades were dest	ut 175 feet above ground level e exited the exhaust pipe. The e troyed, and the tail boom separ	(AGL) when the engine lost po ngine then failed, and the pilot rated from the fuselage.	wer. The pilot said that the conducted an autorotative
June 11, 1985	Juneau, Alaska	Bell UH-1B	substantial	1 fatal
The helicopter lost landing in tall tree last engine overh- torque load. The co condition and brea	t power while transporting a load o s. The report said that the spacer aul. During the accident flight, th ompressor assembly then uncouple akup of the turbine assembly.	f logs over mountainous and de between the no. 5 compressor a e spacer slipped into its prope ed and forced the impeller into its	nsely forested terrain. The pilot and the impeller had not been er position, thus releasing an s magnesium housing. This rest	conducted an autorotative seated properly during the accumulated compressive ulted in an overtemperature
June 25, 1985	Hoquiam, Washington	Bell UH-1B	minor	1 fatal, 1 none
The pilot was flyin	g the helicopter in a hover when th	e external-load sling broke and	I a block of wood struck and kil	led a ground crewmember.
Aug. 8, 1985	St. James, Louisiana	Bell UH-1B	substantial	1 fatal
The helicopter wa the pilot radioed, ' struck the ground fourth-stage rotor	s being flown in a hover at 150 fe Get out of the way." The pilot cond at a high vertical speed. The rep had experienced progressive-fati	et AGL to pick up a log when th ducted an autorotative descent ort said that the engine had se gue failure.	e engine lost power. Witnesse with very low rotor rpm throug vere internal damage. Four co	s heard a "loud bang," and h tall trees. The helicopter ompressor blades from the
March 3, 1986	Bullard, Georgia	Bell 214-B1	substantial	1 minor
The helicopter wa engine had experi The report said th	s lifting a log when a loud explosic enced an uncontained failure of th at the turbine disk had been insta	n was heard and the engine los e first-stage turbine disk. Debris Illed improperly.	st power. The pilot conducted a s damaged the engine cowling	n autorotative landing. The and the main-rotor blades.
March 7, 1986	Siletz, Oregon	Aerospatiale SE-318C	substantial	1 fatal
The pilot tried uns would have increa external load. The when the engine I compressor stall,	successfully to use the helicopter used the helicopter's gross weight pilot was setting the load back or ost power. The helicopter then str and the turbine section had been	to lift a 1,597-pound (724-kilog to approximately 880 pounds the ground, so that some of the ruck the ground with the extern burned because of improper e	gram) load of cedar blocks. Th (399 kilograms) over the maxi ne cedar blocks could be unloa al-load line still attached. The ngine operation by the pilot, th	e report said that the load mum gross weight with an aded by ground personnel, engine had experienced a ne report said.
May 31, 1986	Wenatchee, Washington	Bell UH-1L	destroyed	1 fatal
The helicopter wa released the sling densely wooded a had failed when th	is lifting a sling load when ground load and said that he would land irea. The report said that the 42-do ne input-pinion gear fractured bec	d crewmembers told the pilot the the helicopter. The pilot then I egree gearbox, which had beer ause of corrosion-induced fatig	nat smoke was coming from the ost tail-rotor control and attem overhauled 195 service hours use.	ne tail-rotor area. The pilot pted a run-on landing in a s before the accident flight,
Nov. 13, 1986	Winlock, Washington	Hughes 369D	substantial	none
The helicopter wa the external line of autorotative landir	s being used to load Christmas tricontacted treetops. The sling rebing.	ees onto a truck. The pilot was ounded into the tail rotor and	flying the helicopter to the pick damaged the tail-rotor gearbo	kup area when the sling on bx. The pilot conducted an
April 21, 1987	Bay Minette, Alabama	Bell UH-1B	substantial	none
An internal failure helicopter sank. T	caused the engine to lose power of the engine had accumulated about	during a logging operation. The t 980 service hours after its las	pilot conducted an autorotative st overhaul.	e landing on water, and the
May 13, 1987	Siletz, Oregon	Aerospatiale SA-315D	substantial	1 minor
The report said th during a logging o spun to the groun	at chockers (devices used to pick peration. The external line then be d.	up logs) were not attached pro came entangled with the tail ro	operly to the external line and tor, causing failure of the tail-ro	disconnected from the line otor system. The helicopter

late	Location	Helicopter Type	Helicopter Damage	Injuries
an. 28, 1988 he helicopter w wamp. The repo ying the helicopt Jel-quantity gau	Stockton, Alabama as being flown from a logging si rt said that there was no fuel in t er, which had a larger engine thar ge was inaccurate and that the p	Bell 204B te to a refueling barge when th he fuel tank when the helicopte a similar helicopter operated by ilot identified the fuel-low warning	substantial e engine lost power. The pik was recovered. The pilot wa the company. The report said ng light as a 20-minute light.	1 serious ot landed the helicopter in as relatively inexperienced i that the accident helicopter
uly 5, 1988	Joseph, Oregon	Kaman HH-43	substantial	1 serious
he pilot was ma ee.	neuvering the helicopter into pos	sition for the first external-load p	ickup of the day when the m	ain-rotor blades struck a ta
uly 12, 1988	Walterboro, South Carolina	Bell TH-1L	substantial	1 serious
he 42-degree g ear tooth on a o verhaul procedu elicopter was c elicopter was n ccidents during	earbox failed when the military-s quill pinion had failed because of ures. The helicopter had accum ertified for external-load operatio ot designed for repeated heavy logging operations.	urplus helicopter was lifting a lo f fatigue. The report said that th ulated more than 13,000 lift cy ns by the U.S. Federal Aviation lifting. The report said that sur	ad of logs. The pilot conducto ere was no evidence of pre- cles in 380 hours of service Administration (FAA), but the plus UH-1/TH-1 helicopters	ed an autorotative landing. A existing damage or imprope e in logging operations. Th e manufacturer said that th were involved in two simila
uly 13, 1988	Oakridge, Oregon	Aerospatiale SA-315B	substantial	1 serious
ne pilot lost cor	trol of the helicopter after a log in	n the external-line sling struck a	tree. The helicopter then dea	scended to the ground.
ug. 4, 1988	North Bend, Oregon	Bell UH-1E	substantial	1 minor
uring a logging tor-drive input nce overhaul.	operation, the pilot perceived a tr gear in the 42-degree gearbox ha	ransmission malfunction and co ad failed from fatigue. The milita	nducted an autorotative landi ry-surplus helicopter had acc	ing on rough terrain. The tai cumulated 373 service hour
arch 3, 1989	Forest Grove, Oregon	Schweizer 269C	substantial	1 minor
ne report said th the trailer. The	hat the helicopter was being used helicopter then struck the ground	to load holly trees onto a trailer t d.	ruck when the helicopter's la	nding skid contacted the sid
ay 17, 1989	Ketchikan, Alaska	Bell 214B-1	substantial	2 serious
ne report said ontacted a tree,	that the pilot misjudged clearand and the helicopter struck the gro	ce from trees while conducting bund.	a hover descent to pick up a	a load of logs. The tail roto
ine 14, 1989	Baring, Washington	Oregon Helicopters UH-1L	substantial	1 serious, 1 minor
he helicopter wa he helicopter w olled down a hill duced overstre he U.S. Navy UH	as lifting a log when the crew hea as in a high hover when it begar I. The report said that the input s ss. The engine N <sub>2</sub> (high-pressure I-1 operations manual and the F/	rd unusual noises and felt vibrat n spinning. The pilot conducted piral-bevel gear in the tail-rotor compressor shaft) torque indica AA type certificate specify an N <sub>2</sub>	ons. They released the log and a hovering autorotation. The 42-degree gearbox had failed tor in the accident helicopted torque limit of 50 pounds.	nd flew toward a landing site e helicopter landed hard an d in fatigue because of pilo r had a redline of 55 pound:
ept. 15, 1989	Louisville, Georgia	Sikorsky H-34J	destroyed	none
ne pilot felt a ju ttisoned the ext elieved that the	erking motion while hovering the ernal line and conducted an eme main-rotor-drive free-wheeling u	e helicopter at 200 feet. The m rgency landing in a swampy are nit had failed.	otion occurred several times a. The helicopter burned and	s in a brief period. The pilo was not recovered. The pilo
ov. 15, 1989	Estacada, Oregon	Hughes 269C	destroyed	1 fatal
ne helicopter was a approach to the st control of the hid that the pilot wired to provid lot-in-command	as being used to transport Christ ne drop zone, the pilot did not mai helicopter. The helicopter flew ba had only a manual means of rele e power only to the loading hook d in external-load operations; she	mas trees from a field to a drop intain sufficient altitude or cleara ackwards about 200 feet (61 met easing the external line from the c. The pilot had accumulated fiv did not have the required letter	zone for loading onto a truck nce, and the external load st ers), struck the ground tail-fir helicopter's hook; the electric e flight hours of dual instruct of competency for the accide	. The report said that, durin ruck an obstruction. The pik st and rolled over. The repo cal release system had bee ion and eight flight hours a ent operation.
arch 9, 1990	Cross, South Carolina	Sikorsky HSS-1N	destroyed	1 fatal
ter about one	hour aloft, the pilot flew the helio	copter toward the refueling area	. The engine failed about 20	00 feet over a wooded area
itnesses said t at the helicopte efore the pilot of	hat the helicopter was in a nose- er turned about 180 degrees bef died, he said that the fuel gauge	high attitude and that main-roto ore striking trees. The pilot was was indicating between 100 p	r-blade rotation speed was d removed from the helicopte punds and 200 pounds (45 l	ecreasing. One witness sa r before it burst into flame kilograms and 91 kilogram

Dato	Location	Heliconter Type	Heliconter Damage	Injuries
	Moneks Corpor South		dostrovod	
The pilot flew the manufacture the manufacture the manufacture the manufacture the manufacture the pilot flew t	ne helicopter back to the pick nd released the load. The gr of a main-rotor pillow-block rer had not been followed, an	kup area because the external load ound crew observed the helicopter bolt caused the main rotor to separ nd the pillow-block bolts had becor	of logs was too heavy. He broug pitch nose-up, roll left and desc ate from the mast. Maintenance ne corroded.	ht the helicopter to a hover end to the ground. The repo procedures recommended b
May 21, 1990	Juneau, Alaska	Sikorsky S-58D	substantial	1 serious, 1 none
The pilot lost ta ground. The rep	ail-rotor control while conduc port said that the tail-rotor d	cting a high hover with an external rive-shaft assembly had fractured.	load of logs. The helicopter spu	n right and descended to th
May 21, 1990	Kelso, Washington	Hughes 369D	minor	1 fatal, 1 none
The helicopter ground. One ce personnel clea	was departing from the logg edar block struck and killed a r the area before an aircraft	ing area with an external load of c a ground crewmember. The report departs and that the pilot conduct	edar blocks when the load struck said that the company operating a vertical takeoff to a height clea	tree branches and fell to th manual required that grour ar of obstructions.
July 16, 1990	Ovando, Montana	Sikorsky S-58BT	substantial	none
The pilot was p attempted to fly report said that	reparing to pick up an extern / the helicopter to a suitable t the engine failure was caus	al load when the no. 1 engine lost p landing area, but rotor speed dec sed by a compressor-blade failure.	ower. Flames were seen exiting reased. The pilot then conducted	the exhaust system. The pill an autorotative landing. Th
Aug. 7, 1990	Cougar, Washington	Aerospatiale SA-315B	substantial	1 serious
The helicopter descended ver	was hovering over trees at a tically and struck terrain. The	approximately 200 feet and picking e report said that particles of seala	up a load of logs when the engint material were inside the engine	ne lost power. The helicoptone-speed governor assemble
Nov. 8, 1990	Trout Lake, Washington	Aerospatiale SA-315B	substantial	1 minor
As the pilot rele external line bu or malfunctions	eased a load of logs, the heli it was unable to maintain cor s preceded the accident.	copter drifted sideways and the ext trol of the helicopter, which struck	ernal line became entangled wit the trees and terrain. The pilot sa	n trees. The pilot released th id that no mechanical failure
Nov. 26, 1990	Mulino, Oregon	Hughes 369D	substantial	none
A pick-up devic The pilot said th he tail boom.	e on the external line becar hat he attempted to conduct a	ne entangled with an unidentified o an autorotative landing, but rotor sp	bbject, and the external line cont eed decreased, and the main-ro	acted the main-rotor syster tor blades struck and severe
May 21, 1991	Dog Fish Bay, Alaska	Bell UH-1B	destroyed	1 fatal
The helicopter 1,660 pounds (	struck terrain while departing (753 kilograms) over maximu	g with a log on a 150-foot (46-mete um gross weight.	r) external line. The report said t	hat the helicopter was loade
June 27, 1991	Index, Washington	Sikorsky S-64E	substantial	3 serious, 2 minor
After landing to continued to op helicopter and i refueling site. T	o refuel, the pilot exited the perate at full rpm, and the he inadvertently moved the colle 'he helicopter pitched nose-	e helicopter, and the copilot rema licopter controls were unattended. ective control. The helicopter lifted down, rolled over and struck the g	ined in the helicopter to compl The crew chief, who had become off and turned left. The tail rotor s ound.	ete load reports. The engir a airsick, attempted to exit th truck a trailer parked near th
-eb. 23, 1992	Hobart Bay, Alaska	Bell 214B-1	destroyed	6 fatal, 5 serious
At the pilot's re bassengers an otor, and the t	quest, a 30-foot (nine-meter d was flying the helicopter to ail-rotor 90-degree gearbox	) cable had been attached to the h o the base camp when the flight en separated. The pilot attempted an	elicopter to dissipate static elect countered turbulence. The cable autorotative landing, and the he	ricity. The pilot had picked u became entangled in the ta licopter struck tall trees.
March 6, 1992	Hobart Bay, Alaska	Bell 214B-1	substantial	2 serious
The helicopter he engine lost ailed because	was hovering at approximate power. The pilots conducted of fatigue from insufficient c	ely 200 feet AGL while a log was at an autorotative landing on steep, u ooling.	tached to the external line. The p neven terrain. The report said the	ilots heard a "loud bang," ar at the centrifugal impeller ha
luly 28, 1992	Hayfork, California	Bell UH-1B	substantial	1 minor
The pilot releas waw and rapid lo been installed overheated and	sed a load of logs and was fly oss of rotor speed. The pilot l improperly during maintena d failed.	ing the helicopter back to the picku began an autorotative descent, and nce conducted three days before	p area when he heard a "loud ba I the helicopter struck trees. The the accident; the main-rotor dri	ng" that was followed by a le report said that an O-ring ha ve shaft had lost lubrication
Nov. 10, 1992	Elf Point Camp, Alaska	Bell 206B-3	substantial	none

Appendix Helicopter Accidents During U.S. Helicopter Logging Operations, 1983–1999 (continued)					
Date	Location	Helicopter Type	Helicopter Damage	Injuries	
Nov. 27, 1992 The pilot was rec experienced in su blades struck the	Scotts Mills, Oregon eiving instruction on techniques ich operations. During an appro- tail boom.	Hughes 369HS for conducting operations ach with a load of small log	substantial with external loads of Christma s, the pilot landed the helicopte	none s trees. The instructor was r hard, and the main-rotor	
Feb. 19. 1993	Ketchikan, Alaska	Bell 214B-1	destroved	2 fatal	
Witnesses said th external line. The dual-hydraulic-ser and that the opera	at the helicopter entered uncont external load became entanglec rvo actuator had failed from fatigu ator's records showed regular ov	rolled flight while hovering a l in tree stumps, and the he ue. The report said that the h erloading of its helicopters.	t 200 feet AGL with a load of log licopter struck the ground. The li lelicopter was overloaded by 2,0	is on a 150-foot (46-meter) ower piston rod in the right 00 pounds (907 kilograms)	
May 2, 1993	Copper Harbor, Alaska	Bell 204B	substantial	1 fatal, 1 minor	
The helicopter log the operation req being able to cond rotor and the tail- external-line hook factors that contril	ging operation required a steep of uired the pilot to fly the helicopt duct the operation. The pilot cond rotor 90-degree gearbox separa gross-weight limit. The report a buted to the accident.	descent from the pickup site er "straight down in autorota ducted one descent and whi tted. The report said that th also cited inadequate qualit	at 1,200 feet to the drop site at 7 ation" and that the pilot had exp le flying the helicopter in a hove e helicopter had been operated y control and material defects o	70 feet. The report said that ressed reservations about r over the drop site, the tail I with loads exceeding the of the tail-rotor gearbox as	
Vay 8, 1993	Thorne Bay, Alaska	Bell 214B-1	substantial	2 minor	
The pilot said tha autorotative landin because nuts had	t as he maneuvered the helicop ng was conducted on rough, do been improperly installed in the	ter to lift a log, he heard a l wnhill terrain. The report sa compressor case by mainte	oud noise that was followed by a id that foreign-object damage to enance personnel.	a loss of engine power. An o the engine had occurred	
<i>l</i> lay 12, 1993	Anatone, Washington	Kaman HH-43F	destroyed	1 fatal	
of the helicopter, of fatigue that res HH-43F includes a had approved the 1,814 kilograms)	which descended to the ground ulted from "improper certificatio an 1,100-shaft-horsepower (820- installation of a 1,400-shaft-horse	I. The report said that the n/approval of the helicopter kilowatt) engine and an exte epower (1,044-kilowatt) engi	rransmission-rotor input-gear sh modification by the FAA." The I ernal-load limit of 2,300 pounds ( ne and operation with an externa	aft had fractured because FAA type certificate for the 1,043 kilograms). The FAA I-load limit of 4,000 pounds	
Dct. 18, 1993	Beckwourth, California	Aerospatiale SA-315	substantial	1 serious	
After lifting two lo collective control a servo had caused	gs, the pilot heard a loud "thunł and, thus, was not able to mainta l a hydraulic lock.	that was followed by an use in adequate rotor speed. The speed is a speed with the speed is a speed with the speed is a speed with the speed withe speed with the speed with the speed	incontrolled turn to the right. Th e report said that an internal hyd	e pilot could not lower the Iraulic leak in the collective	
Nov. 9, 1993	Toledo, Washington	Bell 47G3B1	substantial	none	
While lifting an ex during the autoro because of power	ternal load of Christmas trees, th tative landing. The report said t impulses from repeated engine	e pilot heard a "clunk" that v hat the transmission shear compressor stalls.	vas followed by a loss of power. bolts had sheared. The pilot s	The helicopter struck a tree aid that the bolts sheared	
lov. 12, 1993	North Bend, Washington	Hawkins & Powers UH-1	B substantial	1 minor	
The pilot heard lo The pilot released 2-degree interme	ud "grinding" sounds and "bangi the external load of logs and the ediate gearbox pinion gear had s	ng" sounds emanating from e external line, and executed everal fatigue cracks.	the rear of the helicopter before a powered autorotation into tre	he lost directional control. es. The report said that the	
lov. 26, 1993	Sublimity, Oregon	Hiller UH-12D	substantial	none	
he helicopter wa xamined the field	ns transporting trees from a field I before flight but did not see the	to a truck when it collided wires.	with unmarked electrical-transm	ission wires. The pilot had	
ec. 29, 1993	Mobile, Alabama	Kaman HH-43B/F	substantial	none	
fter lifting a load n emergency de escent rate. The ght-main rotor he	of logs, the pilot experienced cor scent toward a remote, tempora report said that company mainte ub.	ntrol problems with the helico ary log heliport. The pilot lo enance personnel had faileo	opter. He released the load and t st control of the helicopter whe I to install a cotter pin that secur	he external line, and began n he attempted to slow its ed the retaining nut on the	
eb. 23, 1994	Seaside, Oregon	Soloy H-23D	substantial	1 minor	
The helicopter wa pilot was unable t because of the pil	s in a hover when the pilot releas o arrest the descent. The helicop ot's improper use of the flight co	ed a sling load of wood and oter struck terrain and rolled ntrols.	increased power. The helicopter over. The report said that the he	began to descend, and the elicopter settled with power	

Date	Location	Helicopter Type	Helicopter Damage	Injuries
pril 1, 1994	Rimrock, Washington	Hughes 369D	substantial	1 minor
he helicopter wa elicopter landed ne helicopter wh	s in a hover at 125 feet AGL and hard and rolled over. The pilot s en the engine failed were not s	d zero forward airspeed when said that the fuel supply had b ufficient for a safe autorotativ	the engine failed. The pilot conc een exhausted. The report said e landing.	ducted an autorotation, and th I that the height and velocity
lay 27, 1994	Libby, Montana	Bell 204B	substantial	1 serious
he helicopter wa mergency desce tuck the terrain.	as being used to lift a log wher ent toward mountainous terrain.	the 42-degree gearbox inpu The report said that the pilot	tt-pinion gear failed. The pilot r was attempting to initiate a lar	eleased the log and began and ing flare when the helicopt
uly 17, 1994	Pilot Rock, Oregon	Bell UH-1B	substantial	1 serious
he pilot said tha nechanical defici	t the helicopter began rotating ency was found during the pos	left and that the anti-torque p taccident inspection.	pedals would not move. The rep	port said that no evidence of
ct. 30, 1994	Willits, California	Kaman HH-43F	substantial	1 serious
he pilot had con utorotative landi ecause of fuel e	npleted logging operations an ng on the side of a hill. The helio xhaustion.	d was returning to the landir copter then rolled down the hi	ng site when the engine flame Il and struck a tree. The report s	d out. The pilot conducted a said that the engine flamed o
)ct. 31, 1994	Willits, California	Bell UH-1E	substantial	1 minor
he helicopter wa anding.	is departing from a drop site wh	en the tail-rotor assembly str	uck a tree and separated. The p	pilot conducted an autorotativ
lov. 19, 1994	Yamhill, Oregon	Fairchild-Hiller FH-1100	substantial	1 serious
he helicopter wa ees. The hook th	is being maneuvered to pick up nen became disentangled and i	a load of Christmas trees wh rebounded into the main-roto	en the external-load hook beca r blades. The helicopter descer	ame entangled with a bundle nded out of control.
eb. 13, 1995	Marblemount, Washington	Sikorsky S-58JT	substantial	1 serious
he main-rotor dr vere not sufficier earbox overhaul	ive shaft failed while the pilot white the pilot white for a successful autorotation. Recommended gearbox overfunction over the second se	vas conducting logging opera The main-rotor drive shaft h naul time and main-rotor drive	tions. The report said that the h ad accumulated 952 service h e shaft retirement time are 2,50	nelicopter's height and veloci ours after installation during 0 service hours.
pril 22, 1995	Walla Walla, Washington	Bell UH-1B	substantial	1 serious
he pilot said tha ilot released the he report said th	t just after the load cleared the load and maneuvered the heli nat the freewheeling one-way c	ground, the helicopter shudde copter for an emergency land lutch assembly had failed.	ered, emitted a loud "scraping" i ding. During the maneuver, the	noise and lost rotor speed. The helicopter collided with tree
une 1, 1995	Umpqua, Oregon	Garlick TH-1L	destroyed	1 fatal
Vitnesses saw th ne. The pilot lost	e tail rotor strike a tree when th directional control, and the he	e helicopter lifted a 3,000-po licopter struck a rock bluff an	und (1,361-kilogram) log using d came to a stop in an inverted	a 150-foot (46-meter) extern I attitude.
ug. 26, 1995	Emigrant Gap, California	Sikorsky CH-54A	substantial	none
Vhile flying upslo egan to conduct elease. The carg puched down an	ope with an external load of log a forced landing, and the cop o hook struck the ground, oper d rolled onto its left side. The re	is, the pilot heard an explosi ilot attempted to release the red and released the logs. Th eport said that compressor st	on. The helicopter pitched nos load. The report said that neit e main-rotor blades struck seve alls had occurred in the right en	e-up and rolled right. The pil her crewmember verified loa eral trees before the helicopt ngine.
ept. 18, 1995	Grace, Idaho	Bell UH-1L	substantial	1 serious
Vhen the helicop he engine's cent irective (AD) rec	ter's military-variant engine los rifugal impeller had separated b uired inspection of the centrifu	t power, the pilot released the ecause of fatigue cracking in o gal impellers in civilian variar	e external load of logs and cond ne of the drain holes. The report its of the engine but not in milit	ducted an autorotative landin said that an FAA airworthines ary variants of the engine.
ov. 9, 1995	Crystal, Michigan	Hughes 369D	substantial	none
he helicopter wa round the ship" a elicopter's fuel-s ater to enter the	as at approximately 50 feet AC as the helicopter descended. Th sump screen contained water a helicopter's fuel system.	GL, lifting a load of trees, wh he helicopter's right skid colla nd that the refueling truck's	en the engine lost power. A w psed during the autorotative la pump filter had been installed	itness observed "black smol nding. The report said that th at an incorrect angle, allowir

Appendix Helicopter Accidents During U.S. Helicopter Logging Operations, 1983–1999 (continued)					
Date	Location	Helicopter Type	Helicopter Damage	Injuries	
ec. 20, 1995	Feather Falls, California	Hughes 369D	destroyed	1 fatal	
he pilot was ma elicopter shake	neuvering the helicopter near a lo and oscillate before it spun to the	ogging site when the sling le ground with the external lir	bad struck trees and terrain near	r a ridge. Witnesses saw the	
pril 4, 1996	Rainier, Washington	Bell UH-1H	substantial	none	
Vhile preparing t nd the pilot cond ccumulated 4,20 rocedures for re	o pick up a load of logs, the pilot h ducted an autorotative landing. Th 2 service hours, including 199 se placing the forward bearing with a	neard several changes in en e report said the ex-military rvice hours in civilian operat an improved forward bearing	gine noise and two "loud bangs." rengine's forward bearing had di ions. Nonmandatory Service Bull g.	The engine then lost power sintegrated. The engine had etin No. 0056 recommended	
uly 13, 1996	Ketchikan, Alaska	Sikorsky CH-54A	destroyed	1 fatal, 1 serious	
he military-surp eard a "popping a an area of cut l allure were inade ailitary-surplus p	lus helicopter was being used to " sound and saw the tail rotor begi ogs. The tail-rotor drive shaft had equate handling/labeling of the rel arts.	lift a load estimated to weig n to slow and the helicopter failed at the no. 5 bearing p ubricated bearing by interm	th 18,000 pounds (8,165 kilogra begin to spin. The helicopter de osition. The report said that poss ediate suppliers, and/or insufficie	ms) when ground personne scended onto sloping terrai ible factors in the drive-shat ent shelf life/service limits fo	
ug. 12, 1996	Coeur d'Alene, Idaho	Garlick UH-1H	substantial	1 serious	
uring external-lo escended out of	bad logging operations, the helico control. The report said that a fai	opter's vertical stabilizer and igue fracture had occurred	d tail rotor separated from the ta in the left side of the forward ver	il boom. The helicopter the tical-stabilizer spar.	
ov. 5, 1996	Leavenworth, Washington	Garlick TH-1L	destroyed	1 fatal	
he pilot was ma ontrol and struck vet hole in the le	neuvering the helicopter from a h k wooded, sloped terrain. Postacc eft spar cap. The report said that t	over to lift a log when the v ident examination of the ve he operator had exceeded	ertical stabilizer separated. The rtical stabilizer disclosed fatigue the design stress limits of the he	helicopter descended out of fractures emanating from licopter.	
ov. 16, 1996	Forks, Washington	Hughes 369D	substantial	1 serious	
he engine lost p ow of fuel throug	ower, and the helicopter struck tr gh the engine fuel nozzles. Conta	ees during the emergency I minants were found in fuel	anding. The report said that fuel samples from the storage tank a	contamination disrupted th nd the helicopter fuel filter.	
ov. 18, 1996	Ballston, Oregon	Bell 206B-3	substantial	1 serious	
litnesses said th as no evidence	hat the helicopter lifted a load of C of airframe or engine malfunction	hristmas trees, began a lef	t turn and then spun to the grour	d. The report said that ther	
lay 22, 1997	Lolo, Montana	Garlick UH-1H	substantial	none	
he pilot said tha ile before landin a hover about escent, and the	t the fuel gauge showed 150 pour g the helicopter at the service are 150 feet above the stump when helicopter touched down hard in	ds (68 kilograms) of fuel rei a, which was about 40 feet ( the engine lost power beca the landing area.	naining, and he decided to trans 12 meters) from the stump. The p use of fuel exhaustion. The pilo	port a large stump to a slas bilot was flying the helicopte t conducted a tight spiraling	
une 3, 1997	Addy, Washington	Garlick UH-1H	substantial	none	
he pilot said tha descent. He the nd the main-roto	t he heard a "high-speed shreddir en heard another loud noise, and or blades severed the tail boom. T	g" sound and felt a vibratior rotor speed began to decre he report said that fatigue o	a during cruise flight. He released ase. The helicopter struck four s cracks were found in the drive sh	the external line and bega mall trees and pitched over aft.	
ug. 27, 1997	Skykomish, Washington	Southern Aero UH-1B	destroyed	1 fatal	
he helicopter sti nain-rotor mast. ,000-hour life lin emoved from UH	ruck the ground after the main-rot The "thin-walled" mast tube had a nit on thin-walled mast tubes in c I-1H helicopters.	or system separated in flig ccumulated 4,007 hours in s ivilian versions of the helico	nt. The report said that a fatigue service. An FAA airworthiness dii opter. The U.S. Army in 1984 ord	fracture had occurred in th rective in 1997 established ered thin-walled mast tube	
ov. 8, 1997	Forest Grove, Oregon	Bell 206B	substantial	none	
he pilot was con ang." The helico ont of the helico ne tail-rotor blac	ducting an approach over a ridge pter turned left, rolled right and st pter, and the main-rotor blades ha le.	to a pickup area in a basin ruck trees and terrain. With d separated from the helico	He said that as the helicopter d esses said that the external line h pter. Cable marks were found on	ecelerated, he heard a "loud nad swung forward and up in one main-rotor blade and on	
ay 19, 1998	Hope, Idaho	Garlick UH-1H	substantial	none	
he pilot said that onstant series of elicopter toward	t he was maneuvering to pick up f compressor stalls." Ground crew a clearing and released the exter	a load of logs when he hea members said that smoke a rnal line. The engine then lo	ard a "weird squealing noise [from nd flames were emitted from the st power, and the helicopter was	m the engine], followed by a tail pipe. The pilot turned the landed hard in the clearing	

# Appendix Helicopter Accidents During U.S. Helicopter Logging Operations, 1983–1999 Date Location Helicopter Type Helicopter Damage Injuries June 18, 1998 Coeur d'Alene, Idaho Garlick UH-1H substantial none The pilot was conducting an approach to drop a load of logs when he heard a "loud metallic bang" from the rear of the helicopter. The engine

then lost power. The pilot released the load and the external line, and landed the helicopter on a disused logging road. The helicopter rocked backward on touchdown and then rolled onto its left side. The report said that the output-reduction gearbox had been damaged by fatigue on the outside of the no. 1 planet-gear shaft. July 9, 1998 Raymond, Washington Hughes 369D 1 fatal. 1 none minor The helicopter was being used to lift a load of cedar-shake blocks when one of the two external-line splices failed, releasing the load. A ground crewmember was struck and killed by debris. The report said that an examination of the other external-line splice showed that the splice did not meet accepted industry standards. July 24, 1998 Libby, Montana Garlick UH-1H substantial 1 serious The helicopter was departing from the pickup area with a load of large tree roots when the roots contacted the ground. The compositematerial external line snapped and rebounded into the tail rotor. The pilot attempted to land in a clearing, but the helicopter struck trees before reaching the clearing. Aug. 10, 1998 Oso, Washington McDonnell Douglas 369D substantial 1 minor The pilot was flying the helicopter at about 250 feet AGL when he heard a "bang" and saw the tail-rotor gearbox go forward past the left side of the helicopter. The helicopter then pitched over and began to turn right rapidly. The pilot conducted an autorotative landing. The report said that the tail-rotor gearbox, tail-rotor output shaft, fork assembly, tail-boom fitting and flex coupling showed signs of overload failure. The tailrotor blades were not recovered. Price, Utah Bell UH-11 Sept. 19, 1998 substantial 1 minor The pilot was hovering the helicopter over a pickup site with a left-quartering crosswind when the helicopter began to turn right. The pilot applied full pressure on the left anti-torque pedal, but the right turn continued. The pilot conducted an autorotative descent. The landing skids struck a beaver dam, the helicopter bounced into the air and pitched over, and the main-rotor blades struck the ground. Oct. 5. 1998 Big Sky, Montana Aerospatiale SA-315B substantial 1 minor A ground crewmember inadvertently attached the helicopter's external line to a log that was still rooted to the ground. When the pilot attempted to lift the log, the helicopter stopped abruptly and pitched backwards. The pilot attempted to release the external line, but the release device (hook) jammed because a clevis (a U-shaped yoke) of improper size had been attached to the hook. The helicopter then struck the ground. Dec. 14, 1998 Darrington, Washington Garlick UH-1H substantial 1 serious The pilot flew the helicopter in a hover while ground crewmembers attached a load of logs to the 150-foot (46-meter) external line. The pilot then was maneuvering the helicopter to remove slack from the external line when he heard a "loud bang." The vertical fin had separated, and the helicopter spun right and descended into the terrain. The report said that all five layers of the left side of the vertical fin had fatigue cracks. March 2, 1999 Shelton, Washington Sikorsky S-64E none 1 fatal, 2 none The helicopter was being used to position a log for delivery at a collection area when one of the two choker lines around the log failed to separate immediately when the long-line release mechanism was operated. One end of the log was lifted back into the air, then fell and struck and killed a ground crewmember who had entered the collection area before both choker lines separated from the long line. South Bend, Washington Hughes 369D June 8, 1999 substantial 1 minor The pilot heard a "loud bang" after releasing an external load and increasing power to depart from the log-collection area. The helicopter began to spin left and descend rapidly. The pilot applied full collective control, and the helicopter landed hard. The report said that a threeinch section of the engine compressor impeller disc had separated from the impeller because of fatigue cracks that propagated from corrosion pits. Sept. 21, 1999 Usk. Washington Garlick UH-1H substantial none The pilot released an external load of logs after losing tail-rotor control. The helicopter then struck terrain. The report said that the tail-rotor 90-degree gearbox mounting studs apparently had sheared. Dec. 3, 1999 Salem, Oregon Hughes 369E substantial 1 minor The pilot was flying the helicopter to the pick-up area at 30 feet to 50 feet AGL when he heard a "bang" and released the external line. The pilot then lost tail-rotor control, and the helicopter began to spin right. The main-rotor blades struck and severed the tail boom during the forced landing. The report said that there was evidence that the tail-rotor blades had struck the external line. Source: Patrick R. Veillette, Ph.D., from reports by the U.S. National Transportation Safety Board



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